MAJOR REPAIR AND ALTERATION

Form Approved OMB No. 2120-0020 11/30/2007

Electronic Tracking Number

U.S Depa Transport Federal A			(A	irfr	ame, Pow	erplan	t, Pı	ropeller, or A	ppliance)		For FAA Use Only		
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and	disposition	of this f	orm. This	repor	ies. See FAR t is required b ederal Aviation	y law (49 t	U.S.C.	ppendix B, and AC . 1421). Failure to i	43.9-1 (or subsequent re eport can result in a civi	evision there penalty not	of) for instructions to exceed \$1,000		
		Nationa N52M	•	egistra	ation Mark				Serial No. 21064870				
1. Aircra	art	Make CESS	NA						Model 210N		Series CENTURION		
		Name (As shown	on re	gistration certi					· ·			
2. Own	er	VOGT	ТНОМ	AS I	M				Address <u>2189 STANF</u> City <u>SAINT PAUL</u>	ORD AVE	State <u>MN</u>		
							Zip <u>55105</u> Country <u>USA</u> 3. For FAA Use Only						
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	ed range fu CFR Part 4				Signature/E 10/15/08			ed Individual B DAUPHINAIS	laghe				
					<u></u>		7. Ap	proval for Return	to Service				
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FAA Fit Standards Manufacturer						Maintenance Orar	aization	De	rson Approved by Canadian partment of Transport				
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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US Department of Transportation Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This form is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

for each	such \	violation (Secti	on 90	1 Federal Aviation Ac	t of 19	958).	r 				
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BY	FAA Inspe	Flt. Standards ector		Manufacturer		Inspection Authorize	ition	Other (Specify)			
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Date of Ap	•	or Rejection 7		Certificate or Designation No. QI2R110L		Signature of Authori	1/	tulk	land	ref	

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements. •

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

CESSNA 210N S/N 21064870, N52MM

REMOVED THE FOLLLOWING:

Apollo 618

INSTALLED THE FOLLOWING:

Bendix/King KLN89B GPS, TSO'ed C129 class A1, originally installed under STC #SA00244WI-D.

Installed the GPS receiver in the radio stack in the location the Loran receiver was removed. Installed the MD41 -528 in the Pilot instrument panel. The GPS steering information is switched onto the pilots HSI and the autopilot left/right input via the MD41-528. All switches and required annunciators are located in the MD41-528. The GPS antenna is located on top of the fuselage above the cockpit.

DOCUMENTATION:

AC43.13 1A chapter 11, paragraphs 424, 429, 443, 450, 519

chapter 15, paragraphs 750, 841, 842

2A chapter 2, paragraphs 21, 22, 23, 27

chapter 3, paragraphs 36, 38(b)

AC20-138 chapter 8 for follow on IFR installation.

Bendix/King KLN89 installation manual p/n 006-10522-0001 Rev. 1, June 1995

STC #SA00244WI-D

MD41-528 installation manual #7016298 rev. 1

Ground and flight tests confirm normal operation and accuracy meets the requirements stated in AC20-138 for follow on IFR installations.

GPS system does not interfere with any other systems.

The maximum continuous electrical load does not exceed 80% of the total rated generator capacity.

A copy of the FAA approved Flight Manual Supplement, dated <u>NAY 0.6 1997</u> was installed in aircraft Flight Manual.

Panel is placarded "GPS NOT APPROVED FOR IFR", pending IFR approval.

Weight and Balance and Equipment list updated. Maintenance records updated. Placed pilot operating handbook in aircraft.

Additional Sheets Are Attached

DEPARTMENT OF TRANSPORTATION AND STATE OF THE PROPERTY OF THE FEDERAL AVIATION ADMINISTRATION

RECEIVED

Form Approved Budget Bureau No. 04-R060.1

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MAJOR REPAIR AND ALTERATION FRA—I EO

(Airframe, Powerplant, Propeller, or Appliance) NOV. 1986 WEU-KOU (Its)

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U.S.Department of Transportation Federal Ariation Administration

MAJOR REPAIR AND ALTERATION. (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

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INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000

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Date 8-30-95								nature of Author	rized Ind	ividual 13cm/	10		•
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Weight and balance or operating limitation changes shall be entared in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Installed Continental IO-550-L 6B engine. Installation in accordance with Bonair Avaition Co. STC SA00274 AT. No weight or balance change.

END

☐ Additional Sheets Are Attached

FAA

DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION

MAJOR REPAIR AND ALTERATION

Form Approved Budget Bureau No. 04-R060.1

FOR FAA USE ONLY

OFFICE IDENTIFICATION (Airframe, Powerplant, Propeller, or Appliance) MSP-1=500 AGC-15 INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) MAKE MODEL CESSNA 210N AKRCRAFT SERIAL NO. NATIONALITY AND REGISTRATION MARK 21064870 USA N-52MM NAME (As shown on registration certificate) ADDRESS (As shown on registration certificate) 2. OWNER 2189 STANFORD AVE. THOMAS H. VOGT ST. PAUL. MN 3. FOR FAA USE ONLY 4. UNIT IDENTIFICATION 5. TYPE UNIT MAKE ATION AIRFRAME POWERPLANT PROPELLER APPLIANCE MANUFACTURER CONFORMITY STATEMENT A. AGENCY'S NAME AND ADDRESS B. KIND OF AGENCY C. CERTIFICATE NO. U.S. CERTIFICATED MECHANIC R.C. AVIONICS FOREIGN CERTIFICATED MECHANIC 294 AIRPORT ROAD CERTIFICATED REPAIR STATION KP5R996M ST. PAUL. MN 55107 MANUFACTURER D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. DATE SIGNATURE OF AUTHORIZED INDIVIDUAL 11/06/91 7. APPROVAL FOR RETURN TO SERVICE Pursuant to the authority given persons specified below, the unit identified in item was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is KAPPROVED REJECTED FAA FLT, STANDARDS OTHER (Specify) MANUFACTURER INSPECTION AUTHORIZATION BY CANADIAN DEPARTMENT
OF TRANSPORT INSPECTOR
OF AIRCRAFT FAA DESIGNEE REPAIR STATION DATE OF APPROVAL OR CERTIFICATE OR SIGNATURE OF AUTHORIZED INDIVIDUAL DESIGNATION NO. 11/06/91 KP5R996M FAA Form 337 (7-67) (8320)

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

R.C. AVIONICS FAA CRS KP5R996M

INSTALLED THE FOLLOWING:

DISCRIPTION

REFERENCE

STEC SYSTEM 60PSS/AUTO TRIM

SA5148SW-D/SA6029SW-D

The above installation conforms to approved methods is AC-43-13-2A where applicable, chapter 2, paragraph 23. section: a, b, c, d, paragraph 27, section: a, b, c, d, e. Ramp and flight checked above installation. installed systems systems perform to manufacturers specifications, and doesn't interfere with other aircraft systems. Added electrical load doesn't exceed aircraft charging system capabilities. Updated aircraft weight and balance to reflect these changes, and made appropriate log book entries.

> ALAN J. AKEE 476641610 11/06/91

ADDITIONAL SHEETS ARE ATTACHED

WU.S. GOVERNMENT RRINTING OFFICE: 1977-771-021/290

FAA Fit. Standards Inspector

FAA Designee

Date of Approval or Rejection JAN. 12.

FAA Form 337 (12-86)

Manufacturer

Repair Station Certificate or Designation No.

KP5R996M

U.S. Department of Forsportation Federal Aviation Administration	i de la company de la comp La company de la company d	MAJOR REP	plant, Pro		pliance)	July .	Office Ident	r FAA Use O	DÔ
and dispositi	on of this form	ype all entries. See F n. This report is requi ection 901 Federal A	red by law (49 U.S.C. 1421). Fa	innie in ieh	Ou cair resour i	ent revision th n a civil penal	nereof) for in ty not to exce	structions in the struction in the struc
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have bee	en made in ac	and/or alteration ma cordance with the re le and correct to the	equirements	i of Part 43 of the U knowledge.	J.S. Federa	ii Aviation Regu	on the revers	e or attachm hat the infor	ents hereto mation
Date	JAN, 1	2, 1990		Signature of Auti	B. Van	Den Br	ocke 47	<u> 476709</u>	2
			7. Ap	proval for Return					
Pursuant to	the authority	y given persons spe eral Aviation Admini	ecified belo	w, the unit identifi	ed in item	4 was inspect	led in the ma	nner prescri	ibed by the
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Inspection Authorization Person Approved by Transport Canada Airworthiness Group

Signature of Authorized Individual Scocke

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets, identify with aircraft nationality and registration mark and date work completed.)

R C AVIONICS FAA CRS KP5R996M

INSTALLED THE FOLLOWING: DESCRIPTION APOLLO 618 LORAN C MISC. ITEMS

REFERENCE APOLLO 560-0037A AC-43-13-2A AC-20-121A

The above installation conforms to approved methods ie AC-43-13-2A where applicable, chapter 2, paragraph 23, section; a, b, c, d, paragraph 27, section; a, b, c, d, e. Ramp and flight checked above installation, installed systems perform to manufacturers specifications, and doesn't interfere with other aircraft systems. Added electrical load doesn't exceed aircraft charging system capabilities. Updated aircraft weight and balance to reflect these changes, and made the appropriate log book entries. The Loran was placarded as "LORAN C Approved for VFR Use ONLY".

Installed an emergency commantenna at F/S 87.00. The antenna was mounted to the aircraft skin which was doubled with a doubler fabricated out of 2024-T3-040 aluminum and riveted to the aircraft skin.

" CURRENT WEIGHT AND BALANCE 01-12-90

CATEGORY NORMAL

EMPTY WEIGHT 2452.95

CENTER OF GRAVITY 42.16

USEFUL LOAD 1347.05

David R. Van Den Brocke 474767092

☐ Additional Sheets Are Attached

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CANADIAN DEPARTMENT
OF TRANSPORT INSPECTOR
OF AIRCRAFT
SIGNATURE OF AUTHORIZED INDIVIDUAL

DATE OF APPROVAL OR REJECTION 01/18/89/

FAA DESIGNEE

CERTIFICATE OR DESIGNATION NO. C14-42

(8320)

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

R.C. AVIONICS FAA CRS C14-42

INSTALLED THE FOLLOWING:

DISCRIPTION

REFERENCE

SIGTRONICS SPA-400 INTERCOM

- SPA-400 MANUAL

The above installation conforms to approved methods ie AC-43-13-2A where applicable, chapter 2, paragraph 23. section; a, b, c, d, paragraph 27, section; a, b, c, d, e. Ramp and flight checked above installation, installed systems perform to manufactures specifications, and doesn't interfere with other aircraft systems. Added electrical load doesn't exceed aircraft charging system capabilities. Aircraft weight & balance was not updated. Weight change of 6oz. is negligible.

ALAN J. AKOE 476641610 01/19/88



ADDITIONAL SHEETS ARE ATTACHED

☆U.S. GOVERNMENT PRINTING OFFICE: 1977-771-021/344

Form Approved O.M.B. No. 2120-0018

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DEPARTMENT OF TRANSPORTATION—FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

21064870 3. AIRCRAFT SERIAL NUMBER Cessna 210 N WANUFACTURER AND MODEL NATIONALITY AND REGISTRATION MARKS

N 52 MM

5. AUTHORITY AND BASIS FOR ISSUANCE
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Normal

None

6 TERMS AND CONDITIONS
Unless sooner surrendered, supprinded, recoved or a termination date is otherwise established by the Administrator, this accordance with Parts 21, 43, and 91 of the Federal Availion Regulations, as appropriate, and the architecture in States.

Hans T. Haenni # DaR-7-FS-EU Nov. 17,1988

Any alleration, reproduction, or musue of this certificate rinay be punishable by a fine not acceeding \$1,000, or movisorment not exceeding \$1,000, or movisorment not exceed the exceeding \$1,000, or movisorment not exceeding \$1,000,

FAA Form 8100-2 (8-82)

AIRWORTHINESS

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PAA AIRCRAFT REGISTRY CAMERA NO. IN DATE: 7-12-89

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for instructi	IONS: Print or type all entries. See FAR ons and disposition of this form.	43.9, FAR 43 App	endix B, and AC 43.9-	1 (or subsequent r	evision thereof)
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	210-64870	트림 하는 데 살되다	N52.4M		
	NAME (As shown on registration certificate)		ADDRESS (As shown of		
2. OWNER	King Aviation Services	Inc.	Love Field Dallas TX 7		-2
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AU.S. GOVERNMENT PRINTING OFFICE: 1977-771-071/290

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Form Approved
Budget Bureau No. 04-R0058

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FAA Form 8130-6 (7-20)

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F. CERTIFICATION—I hereby certify that I am the registered owner (or his agent) of the aircraft described above; that the aircraft is regwith the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal A Regulations; and that the aircraft has been inspected and is airworthy for the flight described. DATE NAME AND ITTLE (Prist or type) SIGNATURE A. Operating Limitations and Markings in Compliance with FAR 91.31 os Applicable B. Current Operating Limitations Affached H. Foreign Airworthiness Certification for Import Aircraft (Attach in required) C. Data, Drawings, Photographs, etc. (Attach when required) L. Previous Airworthiness Certificate Issued in Accordance with FAR CAR (Original attach)			tara magaz Patata	* #45 							
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